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National Aeronautics and  
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# THE ROLE OF A REAL-TIME FLIGHT SUPPORT FACILITY IN FLIGHT RESEARCH PROGRAMS

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## Abstract

This paper presents some of the approaches taken by the NASA Western Aeronautical Test Range (WATR) of Ames Research Center to satisfy the ever-increasing real-time requirements of research projects such as the F-14, F-15, advanced fighter technology integration (AFTI) F-16, YAV-8B, and the X-29A. The approaches include the areas of data acquisition, communications (video and audio), real-time processing and display, data communications, and tracking.

## Nomenclature

AFTI	advanced fighter technology integration
CCD	charge-coupled device
CDI	cockpit display indicator
CID	controlled impact demonstrator
CIMS	calibration and information management system
CRT	cathode-ray tube
DFBW	digital fly-by-wire
FPS-16	precision C-band tracking radar
GPS	global positioning system
HiMAT	highly maneuverable aircraft technology
HUD	heads-up display
MCC	mission control center
MLS	microwave landing system
NALF	Naval Auxiliary Landing Field
NATC	Naval Air Test Center
PCM	pulse code modulated
RPRV	remotely piloted research vehicle
SRV	spin research vehicle
TDRSS	tracking data relay satellite system
TRAPS	telemetry and radar acquisition and processing system
Triplex	L-, S-, and C-band antenna system
WATR	Western Aeronautical Test Range

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## Introduction

In the relatively short time since the first supersonic flight, the amount of data acquired and telemetered to ground-based, real-time experimental support facilities has increased dramatically. The capabilities of these facilities had to be expanded at an even greater rate. The capability of a real-time ground support facility has a direct impact on the cost, both in time and dollars, of flight research.

The primary mission of the NASA Western Aeronautical Test Range (WATR) of Ames Research Center is to provide a capability for the conduct of aeronautical flight research missions through real-time processing and display systems, tracking systems, and communications systems (Fig. 1). The WATR has experienced an almost exponential demand for increased facility capability at its three sites (Fig. 2) at Ames-Moffett, Ames-Dryden, and Crows Landing at the Naval Auxiliary Landing Field (NALF). The mission control center (MCC) or control room (Fig. 3), where the researcher interacts with, interprets, and bases decisions on real-time downlinked and computed data, has become a vital partner in research and flight test. The capability of a real-time support facility to acquire data, perform real-time computations, and display the results in a readily usable and understandable form is critical to the schedule of a research vehicle program. The computational techniques, types of displays, and the ability to "turn around" flight data make the modern real-time ground support facility — with its hardware, software, and staff — a vital member of the experimental research flight test team.

The specific requirements of research and flight test activities vary widely among aeronautical test programs. However, the basic needs are generic to all research aircraft programs: (1) to communicate with the vehicle, pilot, and test team; (2) to acquire, compute, and display real-time data; and (3) to know the exact location of the research vehicle at all times. The never-ending challenge to the real-time ground support facility is to perform these tasks and present the integrated results to the research team in real time (Fig. 4). The evolution of these tasks has seen emphasis placed on different elements of the basic WATR capabilities and need for expansion. This paper presents some of the approaches taken to satisfy the specific requirements of certain types and classes of research aircraft programs.

## Communications

In the formative years of research flight testing, the only means of transmission between the research vehicle and the ground test team was by way of radio communications. The methods of

communications today between the pilot and vehicle and the ground-based test team are limited only by the capability of the facility supporting the research program in real time. The classic method of voice communications using UHF and VHF radio are still used. This method has been augmented by a wide variety of techniques. On programs such as the F-14-IX, the advanced fighter technology integration (AFTI) F-16, and X-29A, the addition of a "hot mike" has proven to be a valuable addition to the test procedure (Fig. 5). The hot mike allows for a continuous flow of pilot comments and observations without requiring the pilot to key a transceiver. This is accomplished by transmitting the pilot's voice, by means of a pulse code modulated (PCM) stream to the ground facility. Within the facility, the PCM stream is reconverted to an analog stream and inserted into the facilities ground intercommunications network and supplied to each researcher. On the X-29A program, this concept has been carried further to include retransmission of the hot mike to safety chase vehicles by means of a voice-activated UHF transmitter. This allows chase pilots access to the same information provided to the ground test team members.

Another approach that has proven valuable to the ground test team is downlinking of the video data from the test vehicle. On the AFTI/F-16, a small charge-coupled device (CCD) camera, located in the heads-up display (HUD), has its output telemetered to the ground. The video signal is fed into the WATR video distribution system and allows the researcher in the MCC to see exactly what the pilot is viewing through the windscreen, as well as what data are being displayed on the HUD (Fig. 6). Other applications of this data acquisition technique include (1) a tail-mounted camera on an F-8 digital fly-by-wire (DFBW); (2) out-the-nose cameras on remotely piloted research vehicles (RPRV) such as the highly maneuverable aircraft technology (HiMAT) vehicle, spin research vehicle (SRV), and B-720 controlled impact demonstrator (CID); and (3) hand-held cameras in chase aircraft for the display of external phenomena that occur on the research vehicle. The upcoming F-18 high angle-of-attack program will utilize three onboard video cameras to downlink tuft patterns and oil flows on the wing and forebody of the vehicle (Fig. 7). The F-18 application will allow researchers in the MCC to view test results in real time instead of waiting until the film has been developed. The F-18 program will also utilize a new approach to the above-mentioned hot-mike technique. The video downlink and will be used as a carrier for the hot-mike downlink and will release the PCM stream, normally used for that purpose, for downlink research data.

Probably the most far-reaching technique being utilized in the WATR is the linking of data in real time to sites across the continent by means of satellite transmission (Fig. 8). This approach has been used on all shuttle landings at Edwards AFB. Harrier data acquired at Edwards AFB have been linked to the Naval Air Test Center (NATC) at Patuxent River, Maryland, by means of satellite to the Wallops Flight Facility and then by microwave land link to NATC. XV-15 tilt rotor

data were linked from the WATR site at Ames-Dryden to the Ames Research Center processing facility at Moffett Field, California. At present, data acquired in real time on the X-29A are transmitted by a secure satellite link to the Grumman Aerospace Corporation facility at Calverton, New York. This approach allows the research team at the Calverton facility to participate in real time in the research flight test program. Additionally, this method of data communication has reduced data turnaround time and has subsequently resulted in a more productive flight test program.

#### Real-Time Processing and Displays

As depicted in Fig. 2, the increased capability required by the types of research aircraft using the WATR facility today have dictated a significant increase in real-time computation and display capability. These types of aircraft can be generically called systems-driven vehicles. They usually have large amounts of data downlinked to the ground test team in real time. The amount of downlinked data is so large that, without real-time computation support, the researcher would have great difficulty assimilating the data for the required real-time decisions in the MCC. To augment the researcher's natural ability, the WATR provides real-time processing support to the MCCs at all sites. The increased number of real-time computations is depicted in Fig. 9. The ordinate value in Fig. 9 is the number of "research" words, in engineering units, delivered to the MCCs. These are words ready for interpretation by the researcher in real time.

The architecture of the telemetry and radar acquisition and processing system (TRAPS) provides a current values table of all computed and downlinked data for access by the displays in the MCCs (Fig. 10). The researcher interface to TRAPS is centered in the WATR MCCs. Two basic types of consoles are available in the MCCs (Figs. 11 and 12). Each MCC at Ames-Dryden is configured with six researcher consoles. Other consoles are available in the MCC for WATR operations, safety and vehicle operations, the flight director, and public affairs. The variety of display devices and the interactive nature of the displays allow for much flexibility for a research program.

The X-29A necessitated expanding the WATR facility to include new displays and procedures that were developed prior to the first flight of the vehicle. Cathode-ray tube (CRT) displays were developed to allow for the display of more than 400 parameters in real time by interaction with the TRAPS. Figure 13 shows several types of displays utilized by the X-29A program. As the program progressed and the envelope expansion continued, it was necessary to provide the researchers with menu selection capability for each stripchart recorder. The researcher was then able to change the parameter lineup for any recorder with a key stroke. The interactive graphics unit shown in Fig. 11 gives the researcher the maximum flexibility. The display allows for 64 different pages of graphics information. Changes can be made in the parameters, scaling, plot type, color, and page layout in real time. Another feature of this console is

the key-stroke callable least squares curve fit for cross-plots. In addition, the key stroke can average certain parameters over a maneuver. The unit can make a hard copy of any page on the CRT, and any page can be distributed to any other console in the MCC.

One of the most cost-effective functions available in the MCC was developed for the F-15 program. This function allows the researcher to uplink computed values to the research vehicle (Fig. 14). The function requires special uplink equipment on the research vehicle and the cockpit displays. When the program is operating in this mode, downlinked data values are used to create computed values that are then uplinked to the vehicle in the form of error signals. The algorithms and vehicle displays are developed in the simulation facility. Prior to its development, the pilot had to perform the function using standard cockpit instrumentation. A conservative estimate of the F-15 project pilot was that, by utilizing this function, the required number of test flights was reduced by 50 percent.

The researcher can interact with the real-time computation to change equation coefficients and parameter values depending on the test point desired for a given maneuver. Given the high cost of flight test time, any function that can optimize the data collected has a significant positive impact on program cost and schedule.

#### Tracking and Data Acquisition

The WATR has an ongoing development program to insure the most accurate downlink data available in real time. High-gain tracking antenna systems and precision tracking radars are used to acquire the optimum data signal. Research vehicles such as the remotely piloted HiMAT (Fig. 15) require continuous uplink and downlink communications at all times at all aircraft altitudes during flight.<sup>1</sup> To accomplish this, two antennas were installed on the vehicle. The PCM downlink was transmitted on two different frequencies, one from each antenna.

This approach takes advantage of the frequency polarization feature of the telemetry autotracking antenna system. Complete radiation coverage is obtained by tuning one receiver to each frequency and using the combing feature of the antenna system. The outputs of the two receivers are combined in proportion to the relative signal strengths to present one PCM bit stream or video output to the ground decommutation station. This approach contributed very much to the success of the flight research program.

One of the most demanding tracking and data acquisition tasks undertaken by the WATR was that required for the B-720 controlled impact demon-

strator (CID).<sup>2</sup> Although no feedback paths from the airplane were necessary to be closed on the ground, precision C-band tracking radar (FPS-16) played a significant role in obtaining the final impact. Manual tracking through visual aids was used to overcome multipath problems. Accuracy could have been improved by using a microwave landing system (MLS). Nevertheless, the final impact was a success, and much valuable information was derived from the impact demonstration. The CID required all WATR tracking and data acquisition systems during each flight mission (Fig. 16). The nature of the project required redundant systems whenever possible.

For precision tracking information in a normal airfield environment, the WATR uses its facilities at Crows Landing in northern California (Fig. 17). The MLS and a laser tracker allow the research vehicle to obtain and use, as required, very precise space position information in a terminal area situation. The integration of these two tracking devices with the X-band radars located at Crows Landing gives the research program the full spectrum of space position information in the MCC.

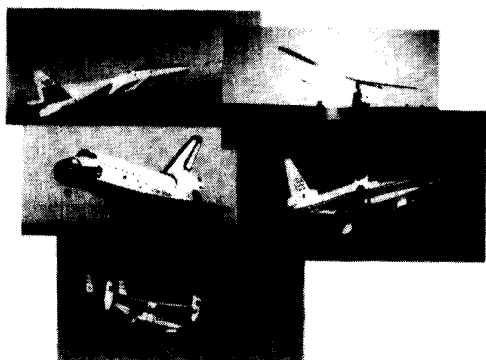
#### Concluding Remarks

The NASA Western Aeronautical Test Range continues to implement approaches and systems to increase the productivity of research flight test. New interactive graphics systems are now under development to enhance the existing capabilities. The installation of new satellite systems will enhance the ability to transmit data from remote test sites to centralized mission control centers (MCCs). The combination of these new satellite systems with existing systems such as the global positioning system (GPS) and the tracking data relay satellite system (TDRSS) will greatly increase the flexibility and productivity of future flight test programs (Fig. 18). The ability to acquire data, track a research vehicle, maintain real-time communications with the research vehicle, and provide comprehensive real-time information in an integrated MCC will insure the close partnership between the research program and the ground-based facility in the future.

#### References

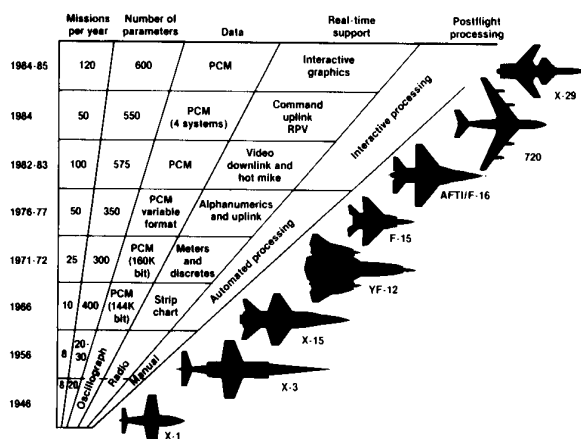
- <sup>1</sup>Harney, Paul F., "Diversity Techniques for Omnidirectional Telemetry Coverage of the HiMAT Research Vehicle," ITC Paper, International Telemetry Conference, San Diego, California, Oct. 13-15, 1981.
- <sup>2</sup>Kempel, Robert W., and Horton, Timothy W., "Flight Test Experience and Controlled Impact of a Large, Four-Engine, Remotely Piloted Airplane," NASA TM-86738, 1985.

Provide capability for the conduct of  
aeronautical flight research through . . .

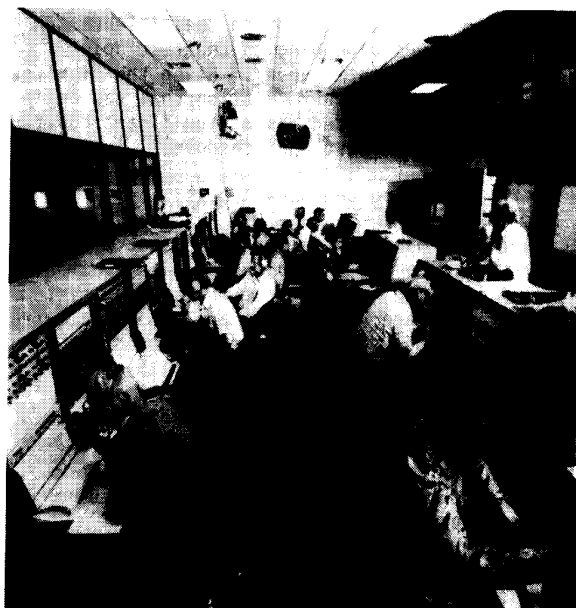


. . . real-time processing and display  
systems, tracking systems, and  
communications systems

*Fig. 1 Western Aeronautical Test Range mission.*



*Fig. 2 Evolution of Ames-Dryden aeronautical  
research program requirements. (For vehicle  
depicted, number of missions per year includes  
flights, combined systems tests, and engine runs.)*



ECN 20640

*Fig. 3 Mission control center at Western  
Aeronautical Test Range.*

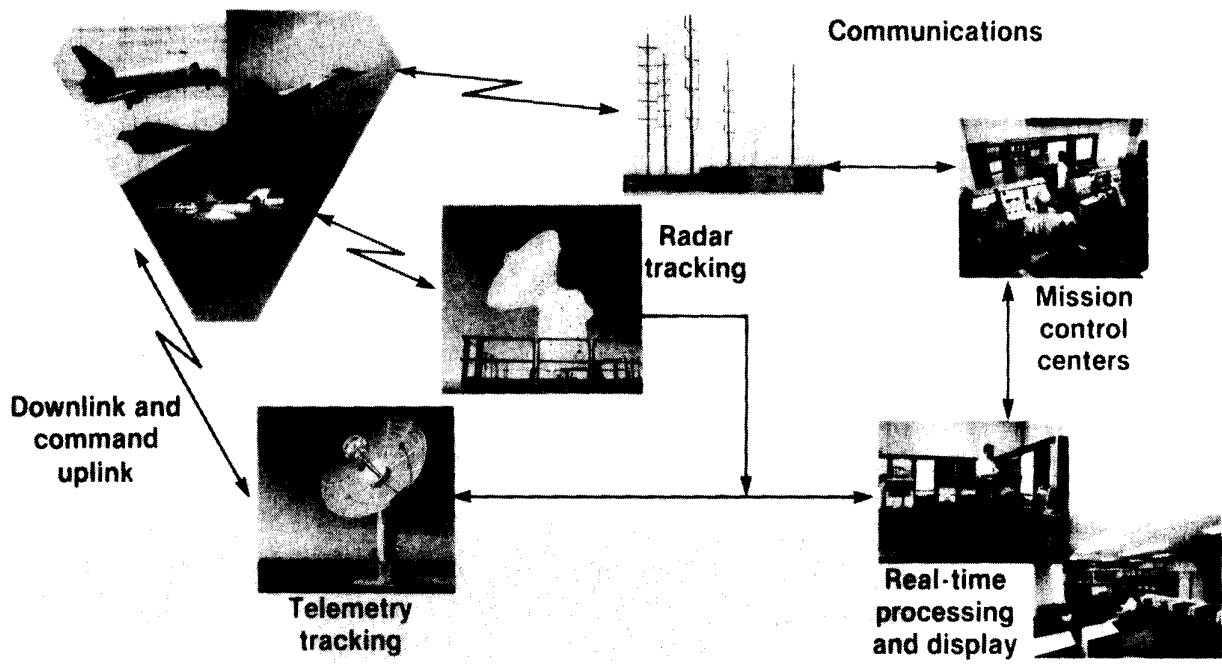


Fig. 4 Basic real-time ground support facility capability for support of research and flight test requirements.

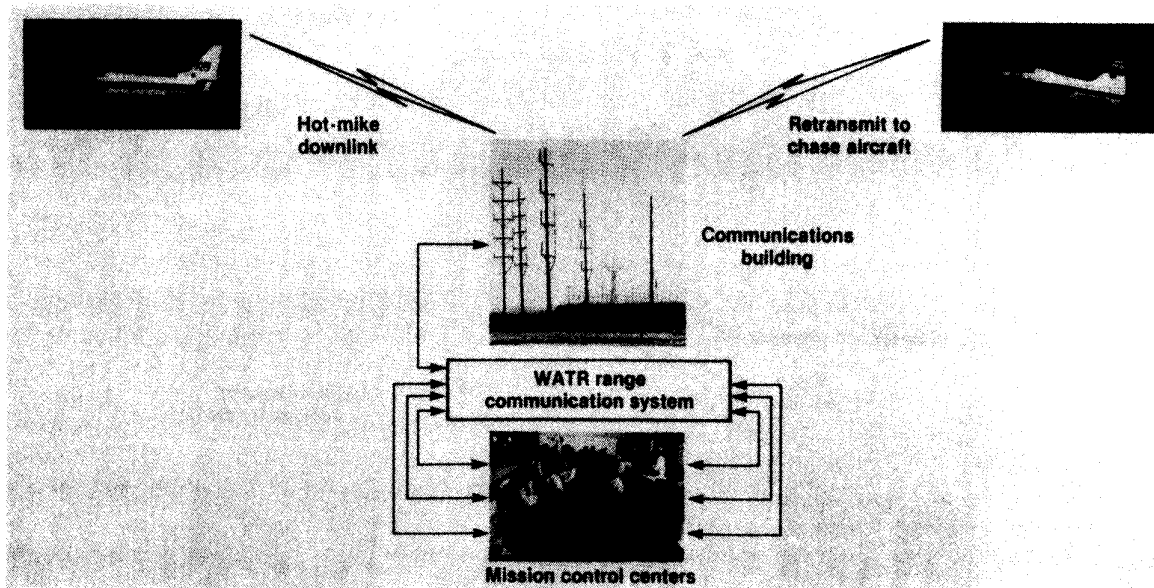
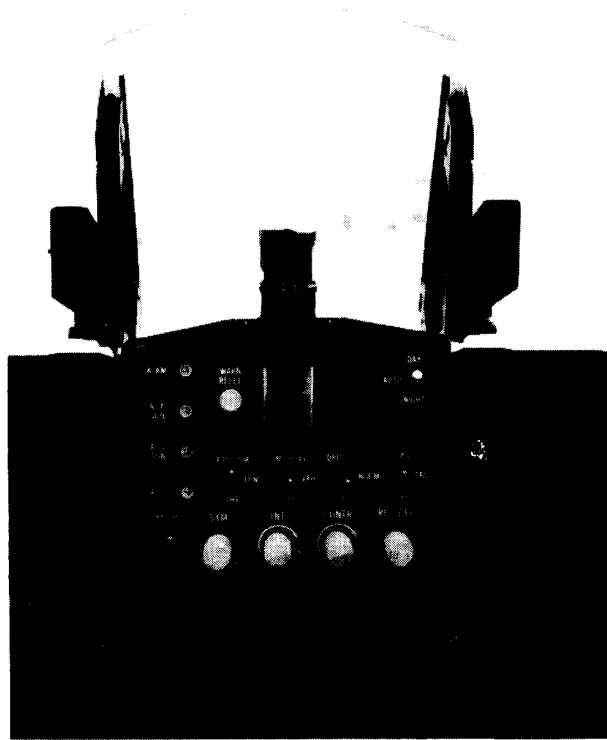
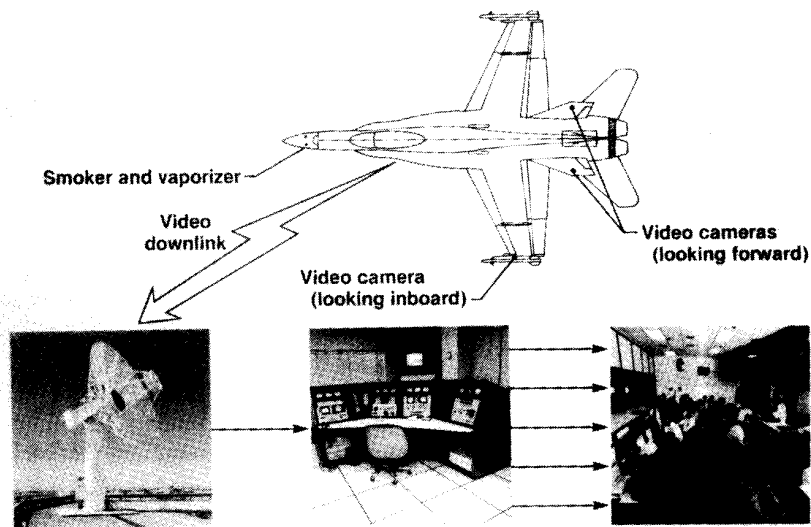


Fig. 5 Typical flight research program and communications network.



ECN 20614

*Fig. 6 Heads-up display.*



*Fig. 7 F-18 real-time flow visualization techniques.*



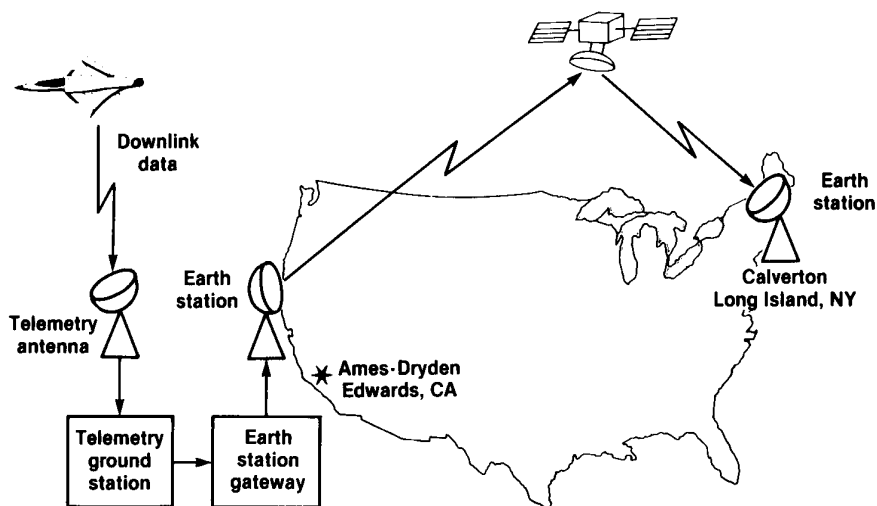


Fig. 8 X-29A transcontinental data link.

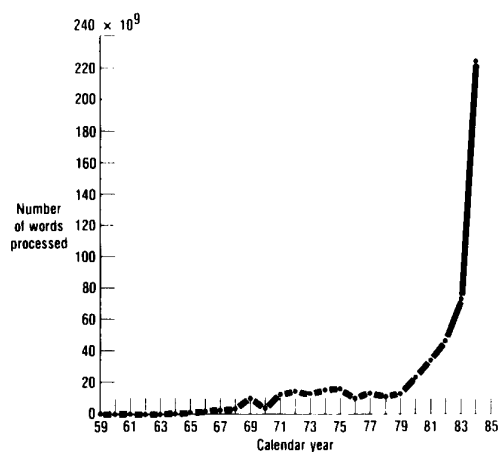


Fig. 9 Total number of real-time computations. (Unified tracking and data acquisition capability attained in 1985.)

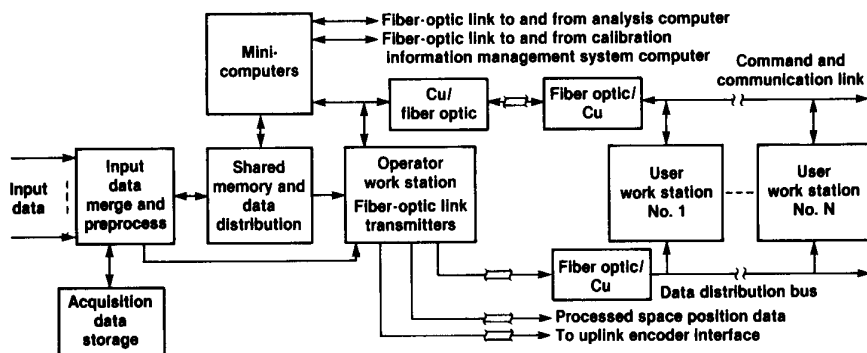
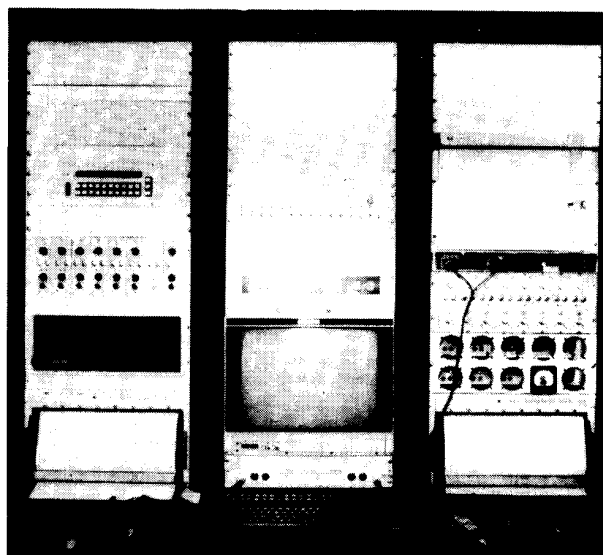
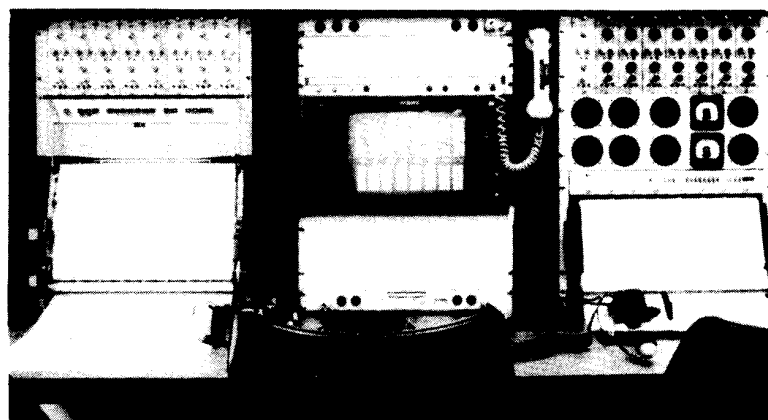


Fig. 10 Real-time data processing and display system.

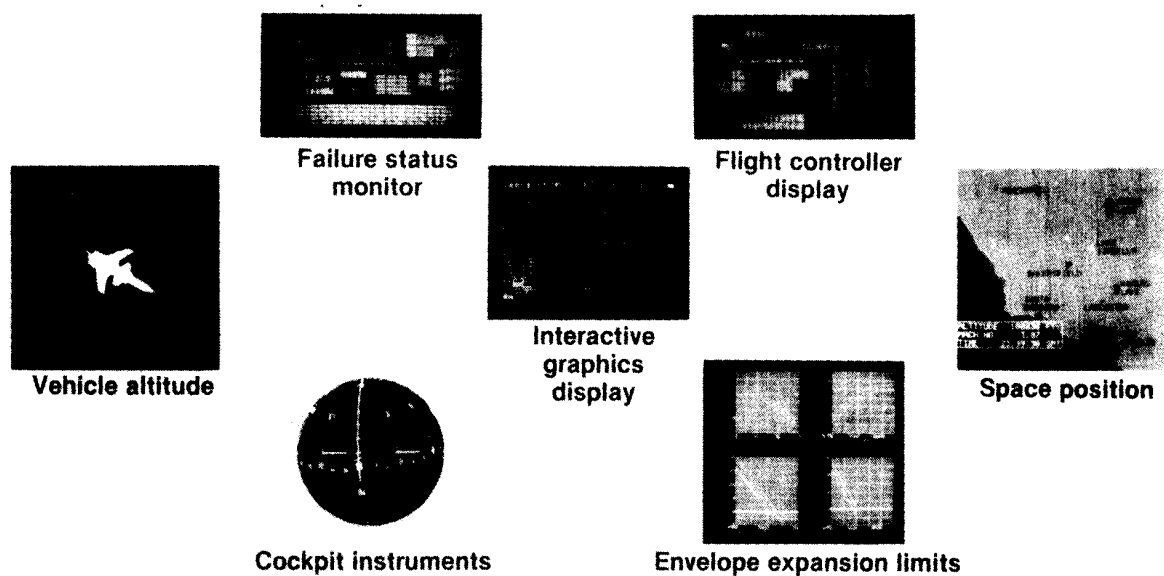


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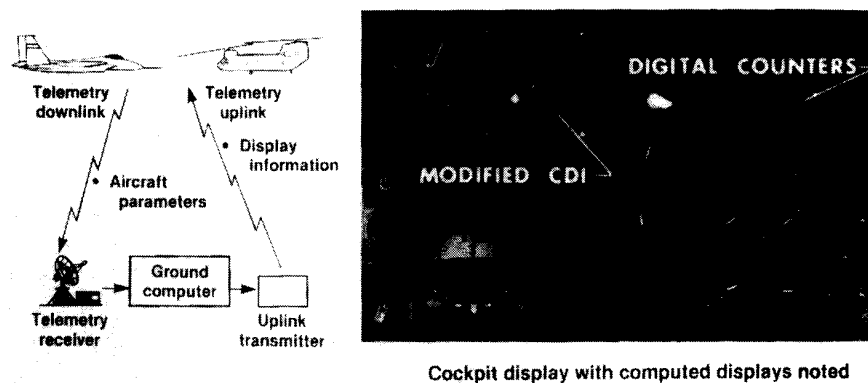
*Fig. 11 Blue room analyst console No. 1 in WATR mission control center.*



*Fig. 12 Blue room analyst console No. 3 in WATR mission control center.*



*Fig. 13 Real-time interactive displays.*



- Ground computer generates display information
- Algorithms developed on simulator

*Fig. 14 Remotely computed display.*

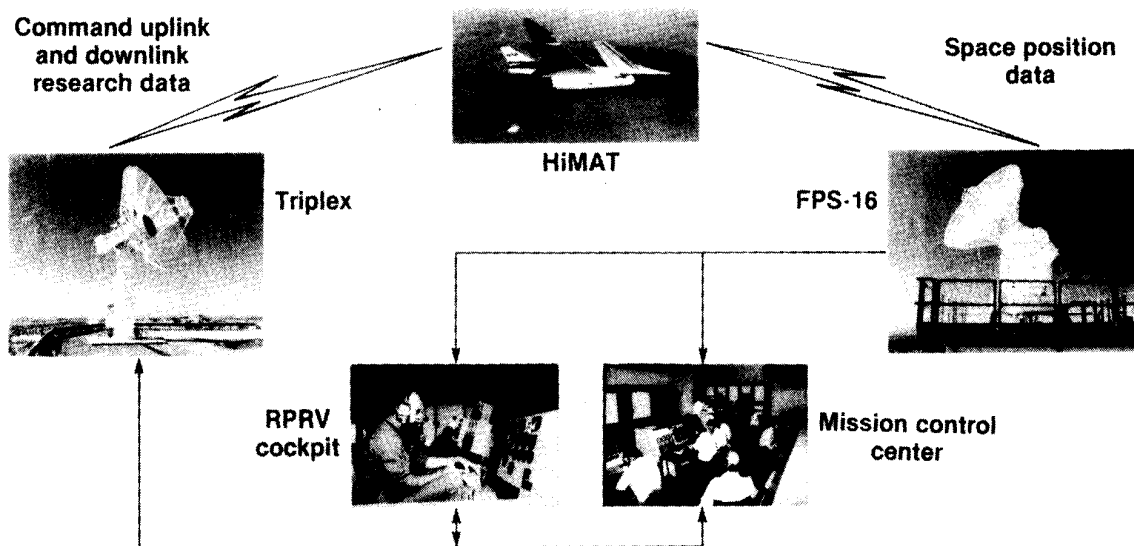


Fig. 15 Tracking and data acquisition requirements for highly maneuverable aircraft technology.

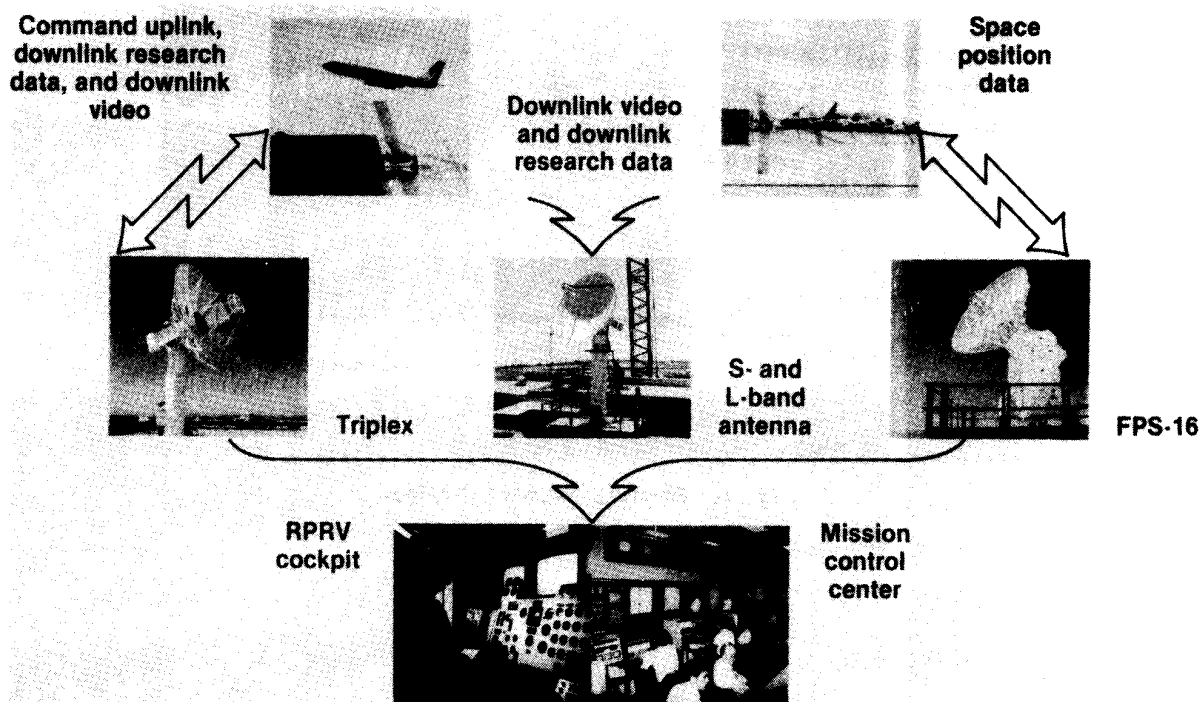


Fig. 16 Tracking and data acquisition requirements for controlled impact demonstrator.

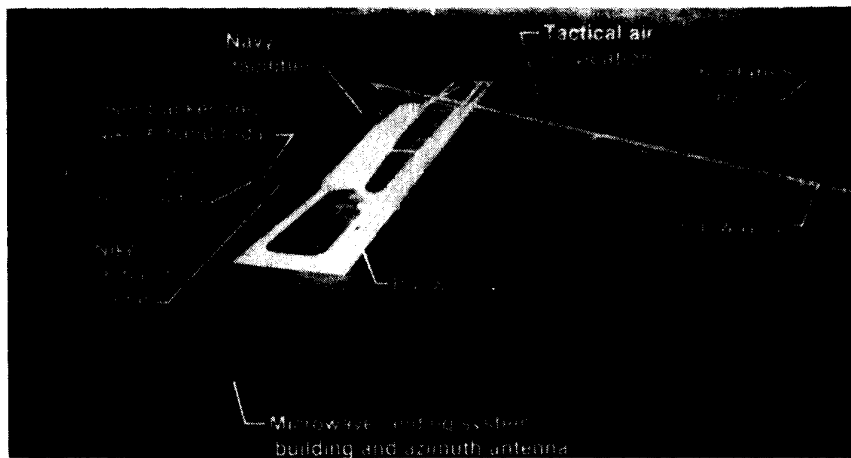


Fig. 17 WATR precision tracking facility at Crows Landing in northern California.

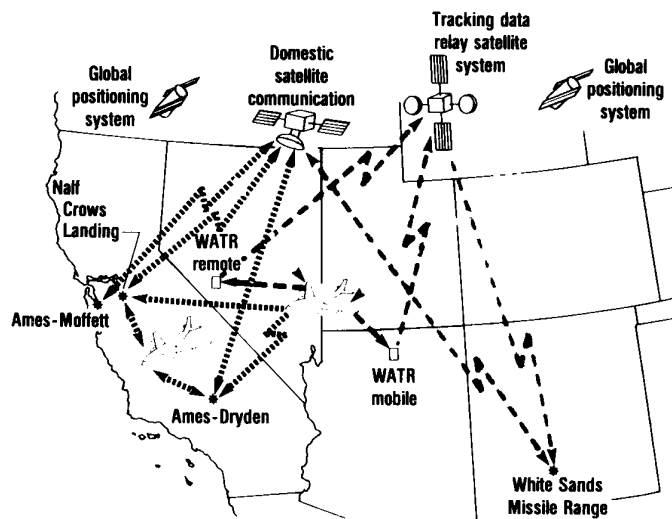


Fig. 18 Future network possibilities.

